

WHY ARE WE HATIN' ON ARTM CPM?

4 KIP TEMPLE

AIR FORCE TEST CENTER EDWARDS AFB, CA

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Kip Temple Air Force Test Center, Edwards AFB CA

ABSTRACT

Why hasn't the Aeronautical Mobile Telemetry community adopted IRIG 106 compliant ARTM CPM as their preferred waveform for the transmission of telemetry data? Telemetry receivers in the market place today exhibit gains in detection efficiency and resynchronization speed that far exceed products of just a few years ago. Past papers have shown the link performance comparison between SOQPSK-TG, the new waveform standard, and ARTM CPM has narrowed since ARTM CPM was first standardized. This paper will present the latest performance comparison between these two waveforms during a controlled test throughout various flight conditions. The flight testing will be presented and performance comparisons are made between the waveforms using traditional methods coupled with several new performance metrics presented in this paper. A comparison of the one true measure of overall link performance, Link Availability is presented for each waveform.

KEY WORDS

ARTM CPM, SOQPSK-TG, IRIG 106, Spectrum Relocation Fund, Link Availability, Data Quality Metric, Trellis Run Length

INTRODUCTION

Of the three waveforms in IRIG 106, PCMFM, SOQPSK-TG and ARTM CPM, only ARTM CPM has failed to gain any level of acceptance by the AMT community. Perhaps this is due to previous work [1] which concluded that even with the excellent spectral occupancy offered by the waveform the initial vendor offerings of receiver/demodulators suffered from synchronization loss at higher values of E_b/N_o , longer resynchronization times, and sensitivity to excessive phase noise when compares to like products for PCMFM and SOQPSK-TG. Follow-on laboratory characterization work in 2013 [2] concluded with the statement "Given the measured performance of current generation ARTM CPM receiver/demodulators, this modulation scheme should be considered a viable modulation scheme for AMT". Still, the AMT community is skeptical. This paper will address the last piece of characterizing the performance of ARTM CPM; performance in a real-world test environment over various types of telemetry channels with a comparison to a baseline of SOQSK-TG. The goal of this paper is to inform the AMT community that ARTM CPM is (and has been) "ready for prime time".

Given the amount of data that was collected and the numerous ways to analyze link performance, there is no way a complete analysis can be presented here. An example analysis is shown to illustrate the process that was followed for the flights and the test points within each flight. An overall comparison of link performance concludes this paper.

THE NEED FOR FLIGHT TESTING

The Spectrum Relocation Fund program at Edwards AFB had one primary goal; test, analyze, and assess the performance of recently installed range upgrades. Secondary to this programmatic requirement was the opportunity to assess and characterize the gains associated with the technologies standardized in IRIG 106. Range upgrades included new antenna feeds, telemetry receivers, receiver status monitoring, and multiband/multimode/coded airborne transmitters. All of these upgrades enabled the Range to first test and subsequently support telemetry systems implementing any combination of Space-Time coding (STC), Low Density Parity Check (LDPC) forward error correction, any IRIG 106 modulation schemes, with signals operating in any of the telemetry bands including lower and mid C-Band.

FLIGHT TEST CONFIGURATION

The flight test program was designed to stress each of the technologies in IRIG 106. The emphasis of the testing was to assess and demonstrate the gains in telemetry link reliably that can be expected when implementing STC and LDPC. A secondary objective, and the subject of this paper, was to assess how recent advancements in receiver technology benefitted ARTM CPM modulation. Expressed another way, has telemetry receiver technology progressed to the point where the selection of modulation scheme isn't a major concern when designing a telemetry link? In order to provide an unbiased assessment, a reference signal was simultaneously transmitted during all of the testing. Though this added equipment complexity in both the aircraft and ground station, it is the only way to provide a direct comparison. By transmitting two signals during each flight, a reference and test signal, many of the normal pitfalls when performing comparison testing during separate flights are negated e.g. differences in flight path, weather conditions, antenna tracking, EIRP, ground station performance (system G/T), etc. To minimize differing transmission channel characteristics due to differences in center frequencies of the two signals, center frequencies were scheduled that followed the recommendation in IRIG 106 for minimum channel spacing.

The aircraft was configured with a transmitter tray that housed two multimode, multiband, coded, STC-enabled transmitters that enabled all of the combinations of test configurations required to simultaneously transmit power level matched reference (REF) and test (TEST) signals. The reference signal for this testing was SOQPSK-TG, the generally accepted baseline for telemetry links in use today. Both of these signals were sent out either a bottom antenna or both a bottom and top antenna depending upon the requirements of the test. The transmitters used internal data and clock with a PRBS-23 data pattern clocked at 5MHz. Since the transmitters were STC-enabled and were using internal data and clock, each could operate as two independent transmitters or as single STC transmitters. The on-board telemetry system recorded time stamped aircraft positional information.

The ground station used for this testing was an SRF-upgraded EAFB range receive site. The intent was to utilize the ground station as if it was supporting a "real" test mission. A 10 foot parabolic dish was used to receive the radio frequency (RF) signals. Left hand (LHCP) and right hand circular polarization (RHCP) are then derived and sent to channel 1 (CH1-LHCP) and channel 2 (CH2-RHCP) of each of four receivers connected to the antenna via a multicoupler.

The receivers were configured to maximal ratio combine CH1 and CH2 with the best channel select option enabled. Up until this point in the receive chain this is a typical telemetry receive station. Where it differed from other ground stations at EAFB was the data sink (control room versus local test equipment) and the added equipment required to capture the necessary data to assess the performance of the TEST and REF links. The data capture equipment consisted simply of two telemetry receiver data loggers, a REACH bit error rate test set with 8-channel capability, two intermediate frequency (IF) recorders, and a GPS-enabled network time server. Additionally the antenna control unit log file was recorded but this did not require a separate piece of test equipment. A block diagram showing the ground station configuration is shown in Figure 1. Each flight the aircraft and ground station equipment captured, recorded, and time-stamped information resulting in the following data products:

- 1. REF Receiver Status log (Channel 1, Channel 2, Combined) @ 1 sample per second
- 2. TEST Receiver Status log (Channel 1, Channel 2, Combined) @ 1 sample per second
- 3. Antenna Control Unit log @ 10 sample per second
- 4. Reach BERT @ 1 sample per second
- 5. Reference Receiver CH1 and CH2 IF Recording
- 6. Test Receiver CH1 and CH2 IF Recording
- 7. Aircraft Positional Information @ 1 sample per second

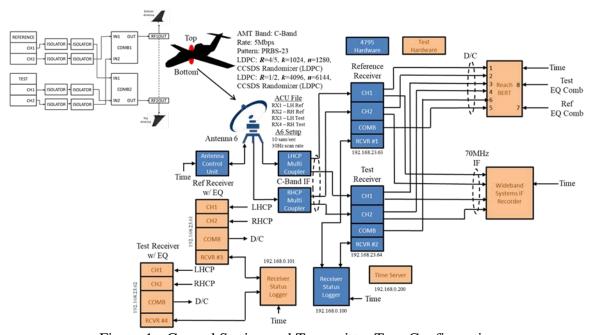


Figure 1 - Ground Station and Transmitter Tray Configuration

Based upon years of flight testing at EAFB, three flight profiles were designed to create three distinct transmission channels: a channel limited by multipath, one limited by noise, and one limited by the composite antenna transmission pattern from the aircraft. The multipath limited channel (points C/D) was created by flying low in both mountainous and flat terrain. This flight path results in a channel that exhibits both long and short delay multipath. The noise limited channel (points H1/H2) was created by flying away from the receive station to the point in which the link is dominated by noise. This profile is flown at a high altitude resulting in higher antenna

elevation angles and is not affected by multipath. The composite antenna pattern channel (points M1/M2 and M5/M6) was created by flying circles at two constant aircraft bank angles in different directions. By doing this, different cuts of the composite antenna pattern are observed by the receive antenna. This point was flown at a higher altitude to mitigate any multipath channel condition as an antenna pattern anomalies are virtually indistinguishable from a multipath event when reviewing the data post flight. These flight profiles along with the test point nomenclature are tabulated in Table 1. The tables show the complete listing of flights that occurred during the SRF-sponsored testing, only data captured and analyzed for Flights 211 and 214 were used for this paper.

Table 1 – Flight Test Configurations

	Ai			
Flight	Reference Signal	Test Signal	Antenna Configuration	Reason for Test
1 (F211)	SOQPSK-TG	ARTM CPM	Bottom Only	Modulation Mode Comparison
2 (F212)	SOQPSK-TG	SOQPSK-STC	Top & Bottom (50/50)	Antenna Pattern Mitigation Assessment
3 (F213)	SOQPSK-TG	SOQPSK-LDPC (R=4/5, k=1024)	Bottom Only	FEC Assessment
4 (F214)	SOQPSK-TG	ARTM CPM	Bottom Only	Finish Flight 1 (F211)
5 (F215)	SOQPSK-TG	SOQPSK-LDPC (R=1/2, k=4096)	Bottom Only	FEC Assessment

Test Points						
Point	Description	Limiting Channel Condition	Flight Conditions			
M1/M2, M5/M6	Antenna Pattern Circle (10°/50° Bank Angle)	Composite Antenna Pattern	13K' MSL, 160 knots			
C/D	Cords Rd (W-E, E-W)	Multipath	5K' MSL, 200 knots			
H1/H2	Isabella/Owens S-N, N-S	Noise	5K'-30K', Best Climb, 160 knots			

TELEMETRY LINK PERFORMANCE METRICS

Historically two performance parameters have been used to characterizing the performance of a telemetry link. First, any combination of receiver signal strength/automatic gain control (AGC) level/signal to noise ratio (SNR) was captured and plotted. Second, bit error data was captured and Link Availability was calculated [4]. This paper will follow that same format. How the receiver is reacting to channel anomalies tells the experienced researcher many things about what is happening with the link. Bit error rate test (BERT) data gives further insight and allows for a Link Availability calculation (Equation 1) which is the one true metric of system level link performance.

$$LA = \left[\frac{\left(T_M - \left(\sum SES + \sum ES + \sum LT \right) \right)}{T_M} \right] (100\%)$$
 Eq. 1

where:

 T_M – measurement period

SES – Severely Errored Second, a one second interval in which the number of bit errors equal or exceed 1×10^{-5} as if these errors were random ES – Errored Second, a one second interval containing at least one bit error but fewer errors than a *Severely Errored Second*

LT – Lost Time, number of bit periods in the measurement period that are

not included in ES or SES

In addition to these traditional link performance metrics antenna pointing error needs to be considered. If the antenna is pointed incorrectly that should not bias the results of a link analysis. Though an incorrectly pointed antenna is an error source, it is not an error source directly attributed to the telemetry transmit/channel/receive chain which are usually under test.

To aid in the link analysis for this paper and in the future, three new metrics are presented in this paper. Two of these metrics are the result of receiver vendor developments coupled with standardization. IRIG 106 Chapter 2 Appendix 2G [3] defines a new real-time link quality metric

appropriately titled Data Quality Metric (DQM). DQM places a numerical value to the quality of a packet of data. The equations that define how the numerical DQM value is determined and scaled is shown in Equation 1. It becomes apparent after reviewing the equations that the key in determining the data quality is an assessment of the bit error probability of the received data.

$$DQM = \frac{-log_{10}(LR)}{k} (2^n) \text{ and } LR = \frac{BEP}{(1-BEP)}$$
 Eq. 2

where: BEP – bit error probability

LR – log-likelihood weighting factor

k – exponent of lowest BEP n – number of DQM bits

(Note: When the source data is known, as is the case for this testing, BEP is a <u>known</u> quantity as BER=BEP. In the general case the source data is not known hence BEP must be estimated.)

Once an estimate of BEP is determined, another very familiar quantity is determined, E_b/N_o . Given BEP and if the modulation method is known (which the receiver <u>must know</u> to demodulate the signal), then an estimate of E_b/N_o can be determined. As an example, most telemetry receivers are capable of detecting SOQPSK-TG at a BER=1x10⁻⁵ for an E_b/N_o ~12dB, thus knowing the estimated BEP and modulation scheme you know the estimated E_b/N_o .

The third new metric presented here is Trellis Run Length (TRL). The IRIG 106 modulation schemes are described by signal states: a sequence of bits maps to a sequence of signal states in a unique way. The transition from one signal state to another signal state may be represented by a state diagram. If time is included, the state diagram unwraps into a trellis diagram [7]. Just as a bit sequence maps to a sequence of states, the same bit sequence maps to a path through the trellis. In a typical receiver, the detection algorithm processes the received signal and attempts to find the path through the trellis (or state sequence) that most closely matches the received signal [8]. At each step in the trellis, the possible trellis paths are compared to the received signal using a quality metric to determine which of the possible trellis paths is the single best path. TRL monitors the metrics to determine the quality of the bit decisions. The lower the corruption (multipath, additive noise, phase noise, etc.) present in the received signal, the greater the number of consecutive symbols have high-quality bit decisions. Higher corruption leads to a smaller number of consecutive symbols with high-quality bit decisions. The TRL metric captures this property and produces a value proportional to the number of consecutive symbols with highquality bit decisions. In this case, a large trellis run length provides higher confidence that symbols are being detected correctly. Captured and recorded trellis run length could be another metric used to evaluate how well a telemetry link was performing at any given time.

FLIGHT TEST DATA ANALYSIS

The flights for the modulation comparison spanned two flights, 211 and 214. Flight paths for these flights are shown in Figure 2. The aircraft system was configured to transmit 5Mbps with a known pattern of PRBS-23 for both SOQPSK-TG (as the REF signal) and ARTM CPM (as the TEST signal) using a single transmitter. These signals were isolated, combined, power matched, and sent out the bottom antenna. The REF signal was centered at 4405.5MHz with an EIRP=+35.6dBm, the TEST signal was centered at 4415.5MHz with an EIRP=+35.4dBm.

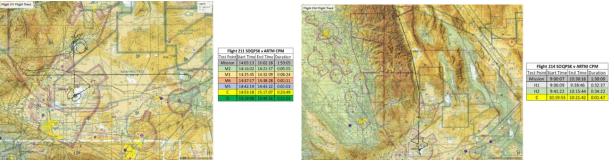
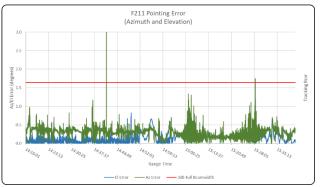


Figure 2 – Flight Paths with Test Point Times, Flight 211

Flight data analysis started with determining antenna pointing error. As previously explained, any bias to the LA calculation by an improperly pointed antenna needs to be removed from the data. Pointing error is determined using the aircraft time-stamped positional data, time-stamped antenna azimuth (Az) and elevation (El) pointing angles, and the rotational center of the antenna in latitude, longitude, and altitude. A calculation is made determining where the antenna should have been pointed. These Az/El angles are then compared with the actual pointing angles. The difference in these angles are the pointing errors. Figure 3 illustrates the resulting pointing error for both Az and El for both flights. There is also an arbitrary reference line plotted for the 3dB full beamwidth for the antenna calculated at the frequency of the test. A close inspection of the plots shows only several instances of Az/El error that would cause a 3dB or more decrease in signal strength. Further investigation of the receiver SNR and BERT files at these times show correlated dips in signal strength but none of these events caused bit errors. Therefore there is no need to remove these times for consideration when calculating LA for both flights.



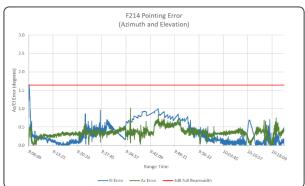


Figure 3 – Antenna Pointing Error

Next, individual test point analysis for the REF and TEST signal was accomplished. Receiver SNR is plotted and evaluated first as it gives an indication as to how the channel is affecting the transmitted signals. Next, estimated DQM and E_b/N_o are compared to the receiver SNR data to verify consistency and to look for any points of interest. Lower SNR values or signal corruption due to multipath should correlate with lower DQM and estimated E_b/N_o and should be consistent between modulation methods. Any points that are not correlated are further investigated. Also plotted along with this data is TRL. This new receiver metric gives an indication on how well the demodulator is making decisions and also give an indication on how hard it is working to make these decisions. Finally, LA calculations are made using the BERT data. By this time in the

analysis process the resulting LA should not be a surprise given the data analyzed before the calculation is made.

Given the amount the testing that was completed it is impossible to cover each test point in this paper. Instead, one test point will be analyzed to illustrate the process that was used and provide a detailed comparison of the two modulations schemes during this test point. Test Point C during Flight 211 is a good point to analyze as it exhibited several interesting channel conditions during the point. First, antenna pointing error during the test point is plotted to again ensure no link errors can be attributed to the antenna. Second, receiver SNR is plotted to illustrate the channel conditions throughout the point and to also identify areas of interest. Figure 4 shows plots of pointing error and estimated SNR of both received polarizations, LHCP and RHCP. Point C (see Figure 2) starts over mountainous terrain, transitions into a flat valley, then ends where line-ofsight is lost at the maximum slant range of the point. With that knowledge the profile of the SNR plot makes perfect sense. At the start of the point SNR is low due to signal blockage by the mountain range. As the aircraft progressed along the test point path it cleared the mountains and descended to 2500' AGL which provided a multipath rich channel with both long and short delay multipath. This is observed at 14:55:19. At 15:00:00 and 15:01:42 there are multipath events which are a well-known, repeatable events along this test point. At 15:14:00 the aircraft flies over a mountain ridge so SNR drops as line-of-sight is gradually lost. Prior to the end there are rapid variations in SNR caused by the numerous mountains in/around the aircraft at that point.

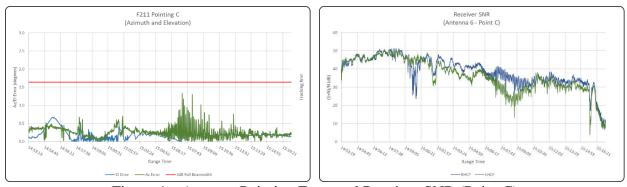
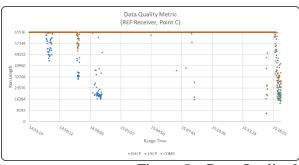


Figure 4 – Antenna Pointing Error and Receiver SNR (Point C)

Generally speaking the DQM and estimated E_b/N_o values track the explained channel anomalies very well. When the signal was corrupted by multipath DQM values dropped accordingly. When signal strength dropped so did DQM. There are differences in how the waveforms were affected by these channel anomaly events though. There are some events where SOQPSK-TG was affected more and vice versa. One possible explanation lies in the occupied spectrum of the waveforms. With less occupied spectrum ARTM CPM is less susceptible to multipath, Figures 4 and 5 verified this. Conversely, if both waveforms were affected by multipath ARTM CPM was more severely affected as more variations in DQM and E_b/N_o are present during these events for ARTM CPM. Notice the DQM of the combiner for both waveforms, it was very happy most of the time. This points to some level of polarization diversity in C-Band.



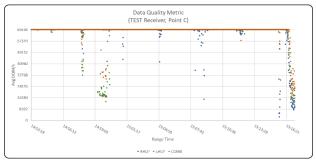
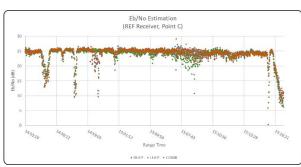


Figure 5 – Data Quality Metric Comparison (Point C)



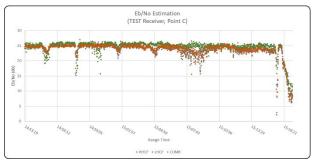
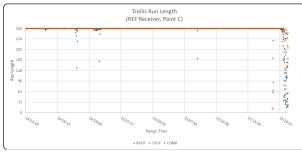


Figure 6 – Estimated E_b/N_o Comparison (Point C)

Trellis Run Length tells us what we already knew, an ARTM CPM demodulator has a harder job of making correct bit decisions than does an SOQPSK-TG demodulator. ARTM CPM is a complex waveform [3] and the variation in TRL between the waveforms tells a story. The SOQPSK-TG demodulator makes higher quality bit decisions more consistently than the ARTM CPM demodulator. Even though there is greater TRL variation for ARTM CPM, the demodulator is still pretty confident on the trellis paths chosen.



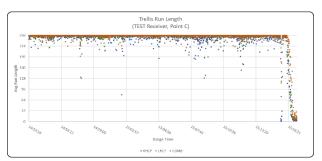


Figure 7 – Trellis Run Length Comparison (Point C)

All of these detailed metrics are very informative and tell a lot about how the signal is corrupted and how that affects the received signal. But what the end user wants to know is: "How good is my real-time data going to be?" When that question is asked, the one overall assessment of link performance used to answer that question is Link Availability. For Point C, LA is shown in Table 3 calculated using BERT data from both the REACH and internal receiver BERT. Given the analysis so far for this test point, these results should be expected. ARTM CPM is on terms with SOQPSK-TG. ARTM CPM has a smaller occupied bandwidth making it less susceptible to multipath than SOQPSK-TG but that is balanced with being a more difficult waveform to detect during disruptive channel conditions.

Table 3 – Link Availability (Point C)

Link Availability												
Flight 211 Receiver BERTs Flight 211 REACH BERT												
	LH	СР	RHCP COMB			LHO	LHCP RHCP			COMB		
Test Point	REF	TEST	REF	TEST	REF	TEST	REF	TEST	REF	TEST	REF	TEST
С	97.7%	97.5%	96.2%	95.7%	99.8%	99.3%	97.5%	97.1%	95.9%	95.5%	99.1%	99.1%

Now that the process for data analysis for one test point has been outlined providing a basis for the Link Availability result, LA results for the entire modulation comparison flights can be presented. For these fights this metric is presented for each polarization, combined output, and test point for both REF and TEST links.

Table 4 – Flight 211 Link Availability

	Link Availability												
Flight 211 Receiver BERTs								Flight 211 REACH BERT					
	LH	LHCP RHCP			СОМВ		LHCP		RHCP		COMB		
Test Point	REF	TEST	REF	TEST	REF	TEST	REF	TEST	REF	TEST	REF	TEST	
Mission	93.7%	93.5%	89.5%	88.2%	98.4%	97.8%	93.6%	92.9%	89.3%	88.2%	98.2%	97.7%	
M2	94.7%	93.2%	88.5%	85.4%	99.7%	99.2%	94.6%	94.4%	86.5%	85.1%	99.4%	99.2%	
M1	98.2%	98.2%	90.9%	89.0%	99.7%	100.0%	98.2%	97.9%	91.7%	89.3%	99.7%	100.0%	
M6	54.9%	54.9%	62.0%	59.2%	83.1%	76.1%	57.7%	56.3%	56.3%	59.2%	83.1%	76.1%	
M5	65.1%	64.8%	60.3%	66.2%	87.3%	83.1%	66.7%	60.3%	58.7%	60.3%	84.1%	82.5%	
С	97.7%	97.5%	96.2%	95.7%	99.8%	99.3%	97.5%	97.1%	95.9%	95.5%	99.1%	98.7%	
D	98.2%	97.2%	98.4%	97.6%	99.8%	99.8%	98.3%	97.5%	98.4%	97.6%	99.9%	99.8%	

Table 5 – Flight 214 Link Availability

Tueste e Tingin 21 (Zink 11 (anderin)									
	Link Availability								
	Flight 211 REACH BERT								
	LHCP RHCP COMB								
Test Point	REF TEST REF TEST REF TEST					TEST			
Mission	95.8%	95.3%	91.9%	90.1%	98.9%	98.4%			
H1	98.6%	98.2%	94.3%	92.2%	99.7%	99.4%			
H2	97.7% 96.8% 96.4% 95.2% 99.2% 98.7%								
C (short)	100.0%	100.0% 100.0% 57.0% 52.3% 100.0% 100.0%							

For Flight 211 LA numbers are derived from both BERTs. Notice the very close correlation between the two results. For Flight 214 only the results derived from the REACH BERT data are presented. This is due to the internal receiver BERT not being configured correctly before flight.

CONCLUSIONS

Most, if not all telemetry transmitters and telemetry receivers currently in use today already have ARTM CPM implemented. The change to a more spectrally efficient waveform for not only spectrally conjected areas but also for every day telemetry mission support is trivial. This paper presented real-world flight test modulation comparison results that support making this change.

Link Availability is the one metric that can be used to compare the system performance of various configurations of a telemetry link. The metric provides an end user an indication on how that certain link would perform at a system level in a real-world test scenario. Other performance metrics (DQM, estimated E_b/N_o , TRL) were presented that can be used to further understand the effects channel anomalies have on the telemetry signal. These metrics help explain the Link Availability results.

WHAT YOU SHOULD GET OUT OF THIS PAPER

• The majority of telemetry links are not noise limited so the difference in detection efficiency (Figure 8) between ARTM CPM and SOQPDK-TG will not be a major contributor to differences in Link Availability. This is verified by the test data presented.



Figure 8 – Eb/No vs BER

- ARTM CPM has a smaller occupied bandwidth making it less susceptible to multipath than SOQPSK-TG but is a more difficult waveform to detect during disruptive channel conditions. These seem to balance when viewed at the system level.
- Link Availability numbers between SOQPSK-TG and ARTM CPM are virtually identical. This is true regardless of how the numbers are compared. On a per mission basis, per test point, per polarization, per combined output, the comparison was favorable.
- ARTM CPM is ready for prime time. Don't be a hater.

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waveform for the transmission of te efficiency and resynchronization sp performance comparison between S CPM was first standardized. This pa controlled test throughout various fi made between the waveforms using	lemetry data? Telemetry receivers in the reed that far exceed products of just a few OQPSK-TG, the new waveform standard aper will present the latest performance cought conditions. The flight testing will be	6 compliant ARTM CPM as their preferred market place today exhibit gains in detection years ago. Past papers have shown the link I, and ARTM CPM has narrowed since ARTM comparison between these two waveforms during a presented and performance comparisons are new performance metrics presented in this paper. ilability is presented for each waveform.

ARTM CPM, SOQPSK-TG, IRIG 106, Spectrum Relocation Fund, Link Availability, Data Quality Metric, Trellis Run Length

17. LIMITATION

OF ABSTRACT

None

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16. SECURITY CLASSIFICATION OF:

b. ABSTRACT

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19a. NAME OF RESPONSIBLE PERSON

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